

Flutter analysis of a hose-drogue system with aerodynamic grid fins for aerial refueling

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Aerial refueling significantly improves the range, endurance, and operational flexibility of military aircraft by allowing mid-flight fuel transfer from a tanker to a receiver aircraft. Among the available configurations, the hose-drogue system remains one of the most widely employed solutions. It consists of a flexible hose stabilized by a drogue, which provides aerodynamic guidance during engagement. However, this configuration is inherently susceptible to aerodynamic instabilities, particularly under turbulent or high-speed conditions, which may lead to severe oscillations, increasing the risk of structural failure and potentially jeopardizing the refueling operation.

This work presents a comprehensive mathematical framework for the analysis of the dynamic and aeroelastic behavior of hose-drogue refueling systems with a new passive stabilization device based on grid fins located at the hose-drogue junction. While the identification of flutter boundaries constitutes a fundamental element of the analysis, the primary contribution of the present study lies in the development and aeroelastic assessment of this novel grid-fin-equipped configuration. Grid fins, composed of a lattice of aerodynamic surfaces arranged within a box-like structure, offer several advantages over conventional planar fins and are introduced to improve system stability.

The modeling approach is based on a formulation capable of predicting both the static equilibrium configuration and the linearized dynamic response of the hose-drogue system. This framework enables an aeroelastic analysis to determine flutter conditions (speeds, frequencies, and modal shapes) across a range of flight conditions and hose configurations.

To quantify the stabilizing effect of the grid fins, in the proposed work a new aerodynamic model is developed to account for both steady and unsteady aerodynamic loads, using the Doublet-Lattice Method combined with a Unit Grid Fin approach. Wind tunnel measurements are used to validate the steady aerodynamic predictions. The grid fins prototype is integrated into the hose-drogue model, allowing for a direct comparison between configurations with and without grid fins. The results demonstrate that the proposed grid-fin configuration leads to a significant increase in stability margins and a noticeable shift of the flutter boundaries toward higher speeds, confirming the effectiveness of grid fins as a passive stabilization solution for hose-drogue aerial refueling systems.