

# CO-DESIGN OF AEROELASTIC SYSTEMS WITH DEEP REINFORCEMENT LEARNING

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## ABSTRACT

### 1 INTRODUCTION

Co-design is a nested or simultaneous optimisation process to optimise both the system and controller. By treating these often highly coupled elements as one, it can reach an optimal solution that a sequential optimisation approach might not achieve [1–3]. One such example is an aeroelastic system navigating through a thermal soaring environment that exploits atmospheric phenomena to extract energy [4–6]. Certain phases of flight require the execution of agile manoeuvres to exploit highly localised thermals, which contradicts directly with an optimal high aspect ratio wing design with high roll inertia and damping. Furthermore, thermal soaring presents an additional challenge of locating thermals and path planning. Deep reinforcement learning has demonstrated remarkable performance in navigating partially observed environments [7–12], and has recently been applied to co-design [13–17]. This motivates the development of a deep reinforcement learning co-design framework for a fixed-wing glider in a thermal soaring environment, shown in Figure 1. In this work, we present 1) a low-fidelity flight simulation with a fully parameterisable elastic high aspect ratio fixed-wing glider model, 2) develop control strategies using deep reinforcement learning for highly flexible aircraft, and 3) integrate data-driven co-design methods in optimising a glider and its control in a unifying model-free framework.

### 2 PROBLEM STATEMENT

A key component of the co-design training process is formulating a well-defined dynamical environment that is also computationally efficient to facilitate reinforcement learning. Here, we developed a flexible aircraft simulation environment in Mujoco [18] by adapting a rigid multi-body implementation of a flexible wing [19]. Mujoco, which stands for multi-joint dynamics with contact, is a physics engine designed to facilitate model and reinforcement learning-based control developments, widely used in the reinforcement learning research community [20]. Many physics extensions have already been developed in Mujoco, such as flapping flight for the simulation of a fruit fly [21]. While many aeroelastic and flight dynamics solvers are available, our goal is to facilitate the use of reinforcement learning for the design and control of flexible aircraft that exhibit aeroelastic phenomena. Therefore, we extend Mujoco and exploit its existing functionalities.

In our proposed rigid panel wing framework, shown in Figures 2, 3 and 4, each wing section is represented by a body with prescribed inertia properties and structural elasticity. The elasticity

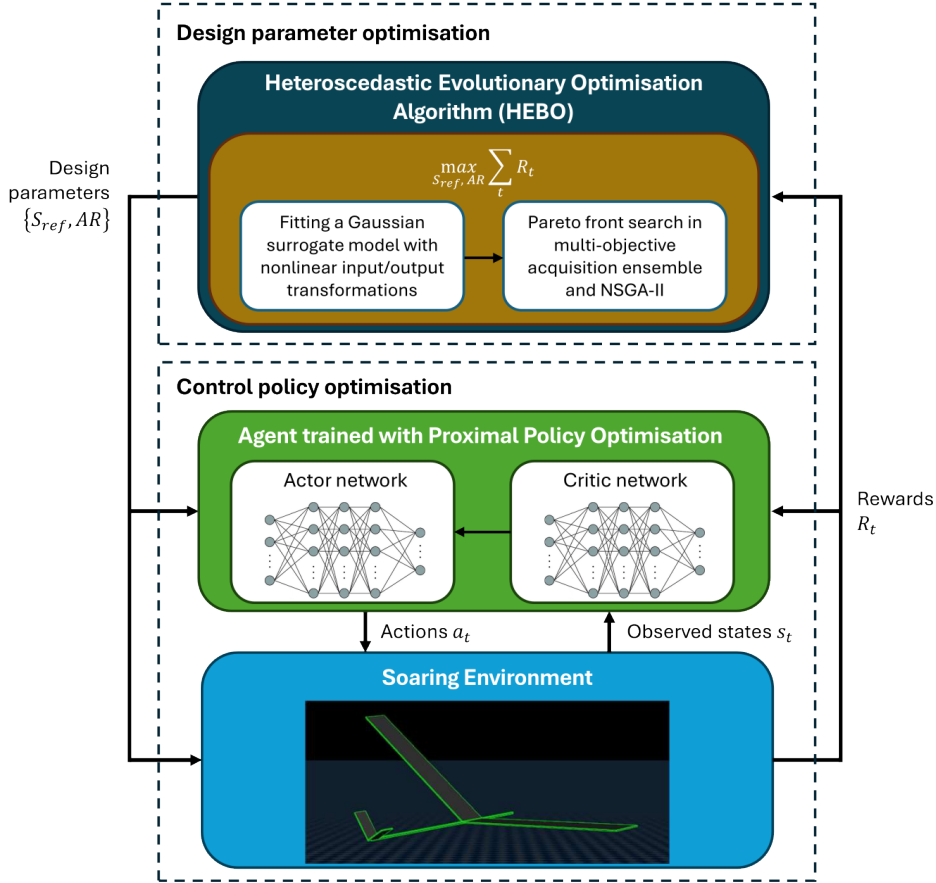


Figure 1: Overview of the co-design process.

is modelled using two torsional springs with linear stiffnesses to represent torsional and out-of-plane bending. A preview of the implementation is shown in Figure 2, where a straight, non-tapered wing is discretised into discrete strip sections. Viterna’s theory was implemented to model the nonlinear lift characteristic [22].

The soaring environment consists of multiple thermal and sink columns, modelled as radial basis functions, as shown in Figure 6. Our nested co-design formulation, depicted in Figure 1, jointly optimises the wing area  $S_{ref}$ , aspect ratio  $AR$ , and its control policy to both control the aircraft and navigate the environment. The objective is to maximise the total reward gained by the agent by maximising the number of mission waypoints reached whilst avoiding ground collision. The agent observes its own state variables, local environment variables, and mission-related parameters, as well as the design sampled from the design space from the outer loop. It

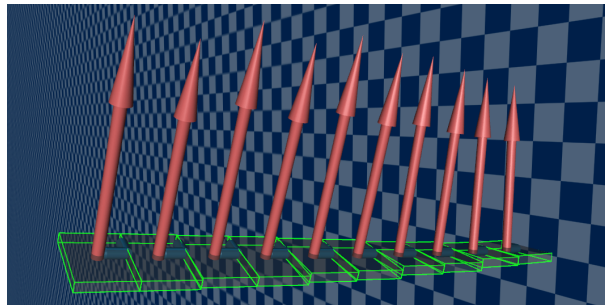


Figure 2: Example discretisation of a clamped port-side wing in Mujoco.

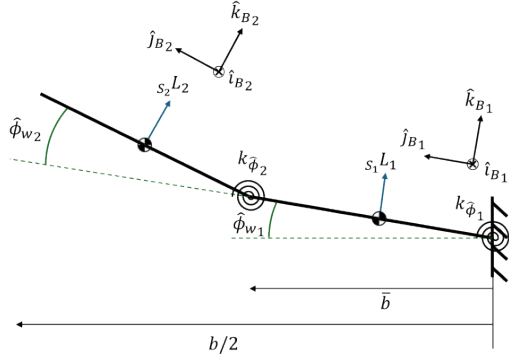


Figure 3: Rear view of a 2 section discretisation of the port side wing.

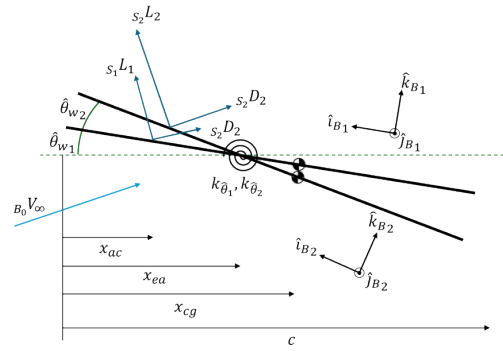


Figure 4: Port side view of a 2 section discretisation.

then interacts with the environment through direct control of the trailing edge flap deflections of its wing sections, as prescribed by its control policy. It receives a tuned reward for its actions, such as reaching a waypoint, which is then recorded by both inner and outer optimisation loops.

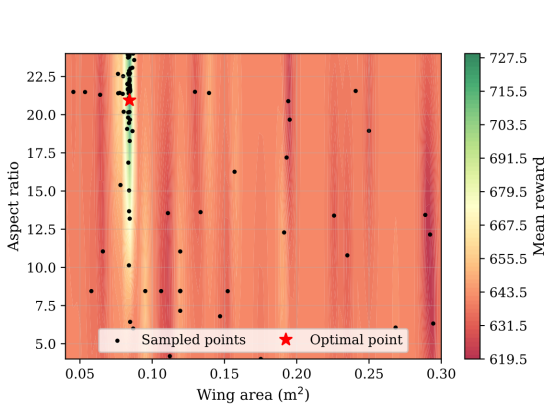


Figure 5: Co-design reward landscape.

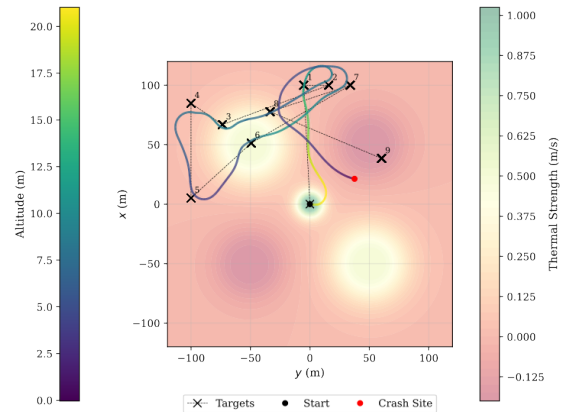


Figure 6: Flight path trajectory undertaken by the advanced agent during evaluation.

The candidate optimal design produced by the proposed formulation is benchmarked against a baseline aircraft optimised solely for its glide ratio. Preliminary studies show that the co-designed aircraft with a 65% smaller wing area and 22% larger aspect ratio has achieved 70% higher mean reward compared to the baseline aircraft. Its reward landscape is shown in Figure 5, and an example trajectory is shown in Figure 6.

### 3 FUTURE WORK

For the full paper submission, we intend to

- Verify the low-fidelity aeroelastic framework and quantify its errors.
- Robustify the optimisation process through design of the agent's observations, actions and its reward structure.
- Explore alternate aeroelastic formulation such as unsteady vortex lattice methods.

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